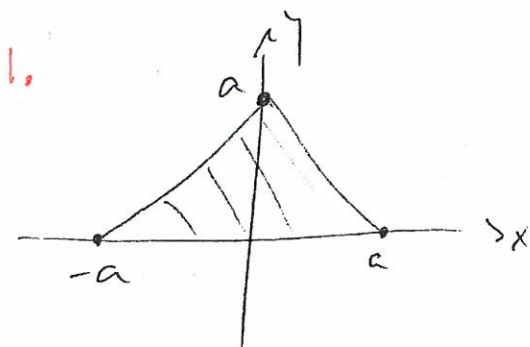


Problem Set 5 Solutions



Uniform mass density ρ

Find \underline{I} wrt origin then translate

$$M = 2 \int_0^a dx \int_0^{a-x} dy \cdot \rho = \frac{1}{2} \rho a^2$$

$$x_{cm} = z_{cm} = 0, \quad y_{cm} = \frac{1}{M} \int_0^a dx \int_0^{a-x} dy \cdot \rho y = \frac{a}{3}$$

$$\bar{I}_{xx} = 2 \int_0^a dx \int_0^{a-x} dy \cdot \rho y^2 = \frac{1}{6} M a^2$$

$$\bar{I}_{yy} = \quad \quad \quad \rho x^2 = \frac{1}{6} M a^2$$

$$\bar{I}_{xy} = \quad \quad \quad \rho xy = 0 = \bar{I}_{xz} = \bar{I}_{yz}$$

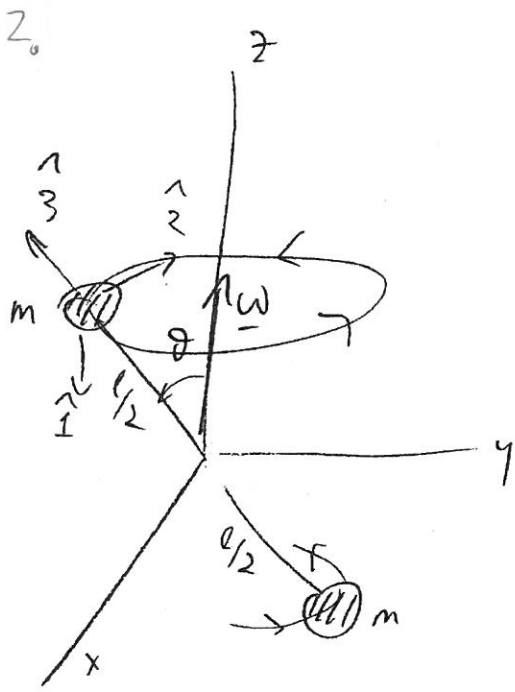
$$\bar{I}_{zz} = \quad \quad \quad \rho (x^2 + y^2) = \frac{1}{3} M a^2$$

Parallel-axis theorem:

$$\underline{\bar{I}}_{cm} = \underline{\bar{I}} - M \left(\underline{r}^2 \underline{1} - \underline{r} \underline{r} \right) \quad \underline{r} = \text{vector from } \underline{0} \text{ to CM}$$

$$= \begin{pmatrix} \frac{1}{6} M a^2 & 0 & 0 \\ 0 & \frac{1}{6} M a^2 & 0 \\ 0 & 0 & \frac{1}{3} M a^2 \end{pmatrix} - M \begin{pmatrix} a^2/9 & 0 & 0 \\ 0 & 0 & a^2/9 \\ 0 & a^2/9 & 0 \end{pmatrix}$$

$$= \begin{pmatrix} \frac{1}{18} M a^2 & 0 & 0 \\ 0 & \frac{1}{6} M a^2 & 0 \\ 0 & 0 & \frac{2}{9} M a^2 \end{pmatrix}$$



$(x, y, z) = \text{space frame (vertical)}$

with $\underline{\omega} = \omega \hat{z}$

$(1, 2, 3) = \text{body frame with}$

$\hat{3} = \text{bar axis}$

$\hat{2} : \text{in plane of } \underline{\omega} \text{ \& } \hat{3}$

$\hat{1} : \perp \text{ plane of "}$

(This choice corresponds to an Euler angle rotation of the z -axis & is convenient.)

Then $\underline{\omega}_{\text{body}} = (\omega \sin \theta, \omega \cos \theta, 0)$

$\underline{I}_{\text{body}} = \text{diag}(I, I, 0)$, $I = 2 \times m \times \left(\frac{l}{2}\right)^2 = \frac{1}{2} m l^2$

$\underline{L}_{\text{body}} = (0, I \omega \sin \theta, 0)$

a) $\underline{N} = \dot{\underline{L}} + \underline{\omega} \times \underline{L}$ in body frame

$N_1 = \omega_2 L_3 - \omega_3 L_2 = -I \omega^2 \sin \theta \cos \theta$

$N_2 = N_3 = 0$

b) Method 1: use the Euler rotation (4.117) for \underline{A}^T

with $\psi = 0$, $\phi = \omega t$ to transfer $\underline{L}_{\text{body}}$ to $\underline{L}_{\text{space}}$

$\rightarrow L_x = -\cos \theta \sin \omega t \cdot L_2$

$L_y = \cos \theta \cos \omega t \cdot L_2$, $L_z = \sin \theta \cdot L_2$

In the space frame,

$$\underline{N} = \frac{d\underline{L}}{dt} = \underbrace{I\omega^2 \sin\theta \cos\theta}_{|\underline{N}|} \underbrace{(-\hat{x} \cos\omega t - \hat{y} \sin\omega t)}_{\text{rotating unit vector in } x-y \text{ plane}}$$

which agrees: the fixed vector \underline{N} in the body frame rotates with frequency ω as seen in the space frame.

Method 2: In space coordinates, the trajectory of the masses m is just

$$\ast \underline{r}_{1,2}(t) = \left(\pm \frac{l}{2} \sin\theta \sin\omega t, \mp \frac{l}{2} \sin\theta \cos\omega t, \pm \frac{l}{2} \cos\theta \right)$$

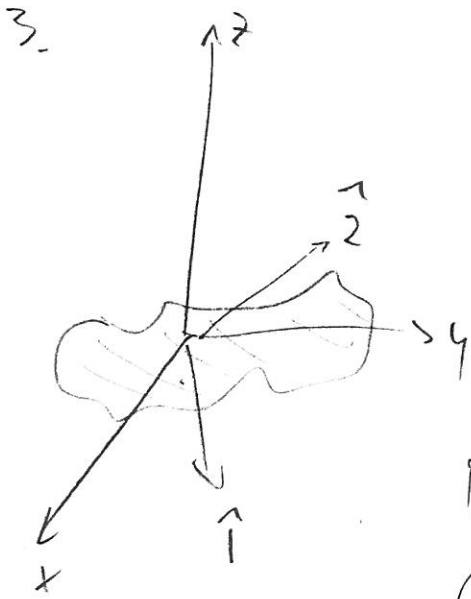
$$\text{so } \underline{L} = \sum_{i=1}^2 \underline{r}_i \times (m\dot{\underline{r}}_i) = \dots$$

$$= \frac{m l^2}{2} \omega \sin\theta \cos\theta \left[-\hat{x} \sin\omega t + \hat{y} \cos\omega t \right]$$

$$\dagger \underline{N} = \dot{\underline{L}} = \text{same thing.}$$

\ast The "phase" of $\underline{r}_{1,2}(t)$ (i.e., choice of $\pm \sin\omega t, \pm \cos\omega t$) is chosen so that at $t=0$,

$$\underline{r}_{1,2}(t=0) \Big|_{\text{body}} = A^T \underline{r}_{1,2}(t=0) \Big|_{\text{space}} = \begin{pmatrix} 0 \\ 0 \\ \pm l \end{pmatrix}$$



Each mass point has $z=0$ so

$$\bar{I} = \begin{pmatrix} \sum m y^2 & -\sum m x y & 0 \\ -\sum m x y & \sum m x^2 & 0 \\ 0 & 0 & \sum m (x^2 + y^2) \end{pmatrix}$$

Diagonalize \bar{I} in the x - y subspace: rotate (\hat{x}, \hat{y}) to $(\hat{1}, \hat{2})$ in the body system, where

$$\bar{I} \Big|_{\text{principal axis}} = \begin{pmatrix} \sum m \hat{y}^2 & & 0 \\ & \sum m \hat{x}^2 & \\ 0 & & \sum m (x^2 + y^2) \end{pmatrix} = \begin{pmatrix} I_1 & & 0 \\ & I_2 & \\ 0 & & I_3 \end{pmatrix}$$

Then note $x^2 + y^2 = \hat{x}^2 + \hat{y}^2$ under rotation, so $I_1 + I_2 = I_3$.

(b) The values $I_1 = \mu^2 - 1$, $I_2 = \mu^2 + 1$, $I_3 = 2\mu^2$ are consistent with (a). The torque-free Euler eqs are

$$I_1 \dot{\omega}_1 = (I_2 - I_3) \omega_2 \omega_3 \rightarrow \dot{\omega}_1 = -\omega_2 \omega_3$$

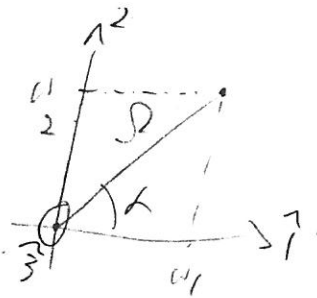
$$I_2 \dot{\omega}_2 = (I_3 - I_1) \omega_3 \omega_1 \rightarrow \dot{\omega}_2 = +\omega_3 \omega_1$$

$$I_3 \dot{\omega}_3 = (I_1 - I_2) \omega_1 \omega_2 \rightarrow \dot{\omega}_3 = \frac{-1}{\mu^2} \omega_1 \omega_2$$

$$\text{so } \frac{d}{dt} (\omega_1^2 + \omega_2^2) = 2(\omega_1 \dot{\omega}_1 + \omega_2 \dot{\omega}_2) = 0$$

$$\text{and } \sqrt{\omega_1^2 + \omega_2^2} = \Omega, \text{ a constant.}$$

(c) let $\tan \alpha = \frac{\omega_2}{\omega_1}$



so $\omega_1 = R \cos \alpha$

$\omega_2 = R \sin \alpha$

+ take $\frac{d}{dt}$:

$$\left(\sec^2 \alpha \right) \frac{d\alpha}{dt} = \frac{\dot{\omega}_2}{\omega_1} - \frac{\dot{\omega}_1 \omega_2}{\omega_1^2} = \omega_3 + \frac{\omega_3 \omega_2^2}{\omega_1^2}$$

$$= \left(\frac{R}{\omega_1} \right)^2 \rightarrow \dot{\alpha} = \left(\frac{\omega_1}{R} \right)^2 \omega_3 \frac{\omega_1^2 + \omega_2^2}{\omega_1^2} = \omega_3$$

Then $\ddot{\alpha} = -\frac{1}{\mu^2} \omega_1 \omega_2 = -\frac{R^2}{\mu^2} \sin \alpha \cos \alpha$

At $t=0$, $\underline{\omega} = \mu N \hat{e}_1 + N \hat{e}_2$ so $\boxed{R = \mu N}$ and

$$\left[\ddot{\alpha} + N^2 \sin \alpha \cos \alpha = 0 \right]$$

To solve this, let $\beta = \dot{\alpha}$ so $\ddot{\alpha} = \frac{d\beta}{dt} = \frac{d\beta}{d\alpha} \frac{d\alpha}{dt} = \beta \beta'$

Integrate: $\int \beta^2 - \frac{N^2}{2} \cos^2 \alpha = C$

+ at $t=0$: $\alpha = \tan^{-1} \frac{\omega_2}{\omega_1} = 0$, $\beta = \omega_3 = N \Rightarrow C=0$

So $\frac{d\alpha}{dt} = N \cos \alpha$ or $\frac{1}{2} \log \frac{1 + \sin \alpha}{1 - \sin \alpha} = Nt + C_1$

so $\sin \alpha = \tanh Nt$, $\cos \alpha = \operatorname{sech} Nt$

and $\underline{\omega}(t) = (\mu N \operatorname{sech} Nt) \hat{e}_1 + (\mu N \tanh Nt) \hat{e}_2 + (N \operatorname{sech} Nt) \hat{e}_3$