

Non-inertial reference frames (Chapter 9)

Started this course by saying that "experiment" \Rightarrow there exist inertial ref frames where Newton's laws apply.

Obvious non-inertial frame - moving train

	train	ground
accelerates	you're pushed backwards	engine accelerates train, train pushes you
turns	you're pushed to side \uparrow extra force	train pushes you to provide centripetal force \uparrow usual $F=ma$

Start in the inertial frame, change coordinate system to moving frame \rightarrow find extra terms = "fictitious forces"

Linearly accelerating frames:

suppose $\underline{F} = m\underline{\ddot{r}}_0$ in inertial frame S_0

If frame S moves at velocity \underline{V} + acceleration \underline{A} wrt S_0 ,

$$\text{then } \underline{r}_0 = \underline{r} + \underline{\Delta}(t), \quad \underline{\dot{r}}_0 = \underline{\dot{r}} + \underline{V}, \quad \underline{\ddot{r}}_0 = \underline{\ddot{r}} + \underline{A}$$

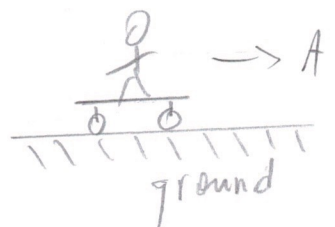
\uparrow
coord in S

$$\rightarrow m\underline{\ddot{r}} = \underline{F} - m\underline{A}$$

$\underbrace{\hspace{2cm}}_{\text{extra "inertial force" seen in } S}$

In particular: if $A = \text{constant}$ this is indistinguishable from gravity (at strength A) - form of "Principle of Equivalence"

Example 1 - moving train



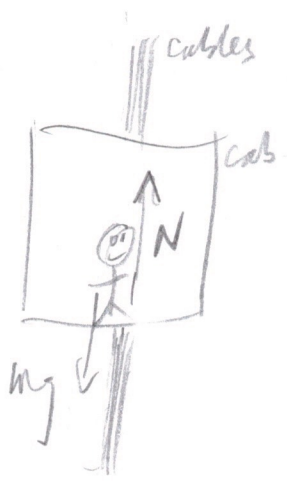
ground frame: $A > 0, x_0 = \frac{1}{2} At^2$

$F = mA = \text{force applied to you to accelerate}$

train frame: $\ddot{x} = 0 = \ddot{x}_0 = F - mA$
 ↑ applied force - friction from train floor ← inertial force

The inertial force pushes back & keeps you at rest in the car (pushes forward if decelerating)

Example 2 - elevator



ground frame $m\ddot{z}_0 = N - mg$

elevator frame $m\ddot{z}' = N - mg - mA \rightarrow 0$
 $A = \ddot{z}_0$
 $= N - m(g + \ddot{z}_0)$

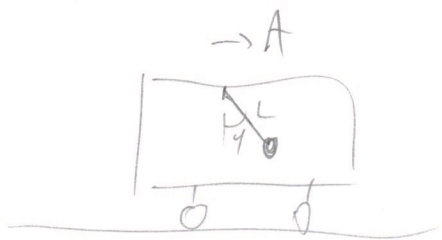
$\ddot{z}_0 = 0 \quad N = mg =$

$\ddot{z}_0 > 0 \quad N = m(g + |\ddot{z}_0|) \quad \text{"extra g"}$

$\ddot{z}_0 < 0 \quad N = m(g - |\ddot{z}_0|) \quad \text{"less g"}$

free fall: $\ddot{z}_0 = -g \quad N = 0$

Ex 3: conic - pendulum in moving train



$$m \underline{\ddot{r}} = \underline{T} + m \underline{g} - m \underline{A} \quad \text{train frame}$$

$$= \underline{T} + m \underline{g}_{\text{eff}} \quad \underline{g}_{\text{eff}} = \underline{g} - \underline{A}$$

(1) looks like ordinary pendulum with g_{eff} at an angle,
 so oscillate about $\phi = \alpha = \tan^{-1} \frac{A}{g}$
 with period $\sqrt{\frac{g_{\text{eff}}}{L}}$



(2) Lagrangian method:

grand frame } $\underline{r} = (L \cos \phi + v, L \sin \phi)$

$$v = \int_0^t dt' v(t') = \int_0^t dt' \int_0^{t'} dt'' A(t'')$$

$$\dot{\underline{r}} = (L \dot{\phi} \cos \phi + v, -L \dot{\phi} \sin \phi)$$

$$L = \frac{m}{2} \left[(L \dot{\phi} \cos \phi + v)^2 + (-L \dot{\phi} \sin \phi)^2 \right] + m g L \cos \phi$$

$$= \frac{m}{2} \left[L^2 \dot{\phi}^2 + 2L \dot{\phi} v \cos \phi + v^2 \right] + m g L \cos \phi$$

$$\frac{d}{dt} \frac{\partial L}{\partial \dot{\phi}} = \frac{\partial L}{\partial \phi} \rightarrow \frac{d}{dt} (m L^2 \dot{\phi} + m L v \cos \phi) =$$

$$= m L^2 \ddot{\phi} + m L \dot{v} \cos \phi - m L v \dot{\phi} \sin \phi$$

$$= -m L \dot{\phi} v \sin \phi - m g L \sin \phi$$

$$or \quad mL^2 \ddot{\varphi} = -mL\dot{V} \cos\varphi - mgl \sin\varphi$$

$$or \quad \ddot{\varphi} = -\frac{g}{L} \sin\varphi - \frac{A}{L} \cos\varphi$$

\uparrow usual problem \nwarrow new term

for design $g = \cos\alpha \cdot g_{eff}$, $A = -\sin\alpha \cdot g_{eff}$

$$or \quad \ddot{\varphi} = -\frac{g_{eff}}{L} (\sin\varphi \cos\alpha - \cos\varphi \sin\alpha)$$

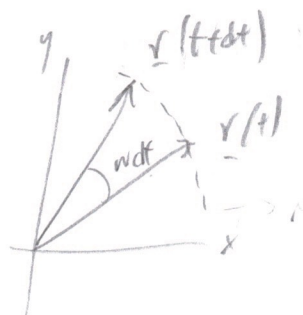
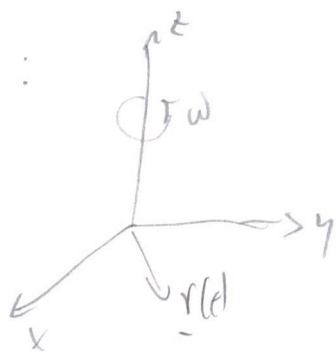
$$= -\frac{g_{eff}}{L} \sin(\varphi - \alpha)$$

\therefore oscillates with $\omega = \sqrt{g_{eff}/L}$ about $\varphi = \alpha$!

— x —

Rotating reference frames:

reminder:



$$\underline{r} = (r \cos \varphi, r \sin \varphi, 0) \xrightarrow{dt} (r \cos(\varphi + \omega dt), r \sin(\varphi + \omega dt), 0)$$

$$= r \left(\cos \varphi \cos \omega dt, -\sin \varphi \sin \omega dt, \right. \\ \left. \sin \varphi \cos \omega dt + \cos \varphi \sin \omega dt, 0 \right)$$

$$\approx r \left(\cos \varphi - \omega dt \sin \varphi, \sin \varphi + \omega dt \cos \varphi, 0 \right)$$

$$= \underline{r}(t) + d\underline{r}$$

Notice $\underline{\omega} dt \times \underline{r}(t) = \begin{vmatrix} \hat{x} & \hat{y} & \hat{z} \\ 0 & 0 & \omega dt \\ r \cos \varphi & r \sin \varphi & 0 \end{vmatrix}$

$$= \hat{x} (-r \sin \varphi \cdot \omega dt) + \hat{y} (r \cos \varphi \cdot \omega dt)$$

$$= d\underline{r}$$

$$\text{or } \frac{d\underline{r}}{dt} = \underline{\omega} \times \underline{r}(t)$$

If $\underline{Q} = Q_x \hat{x} + Q_y \hat{y} + Q_z \hat{z} = \text{vector of fixed length } |\underline{Q}|$

which rotates at $\underline{\omega}$, then $Q_i = \text{fixed}$ but \hat{x} & \hat{y} rotate as \underline{r} above

$$\begin{aligned} \frac{d\underline{Q}}{dt} \Big|_{rot} &= \underline{Q}_x \underline{\omega} dt \times \hat{x} + \underline{Q}_y \underline{\omega} dt \times \hat{y} + 0 \\ &= \underline{\omega} dt \times \underline{Q} \end{aligned}$$

This is completely general - only special assumption was to choose the \hat{z} axis along $\underline{\omega}$ - always possible.

NB - $d\underline{Q}$ = change in \underline{Q} due to rotation = change in the separate components in the old frame.

\underline{Q} is still the vector which was fixed here.

In particular,

$$(\underline{Q} + d\underline{Q})^2 = \underline{Q}^2 + 2\underline{Q} \cdot (\underline{\omega} dt \times \underline{Q}) + o(dt^2)$$

Now suppose \underline{Q} is attached to some body frame S that rotates at $\underline{\omega}$ wrt the original inertial frame S_0 , but allow \underline{Q} to change in S . Then

$$\frac{d\underline{Q}}{dt} \Big|_{S_0} = \frac{d\underline{Q}}{dt} \Big|_S + \frac{d\underline{Q}}{dt} \Big|_{rot}$$

$$\text{or } \frac{d\underline{Q}}{dt} \Big|_{S_0} = \frac{d\underline{Q}}{dt} \Big|_S + \underline{\omega} \times \underline{Q}$$

$$(4) \quad \frac{d}{dt} \Big|_{S_0} = \frac{d}{dt} \Big|_S + \underline{\omega} \times \quad (\text{for vectors})$$

Another way of saying this:

$$\underline{Q}(t) = \sum_{i=1}^3 Q_i(t) \hat{e}_i(t) \quad \hat{e}_i(t) = \text{unit vectors in } \underline{S}$$

$$\begin{aligned} \text{then } \frac{d\underline{Q}}{dt} \Big|_{S_0} &= \sum_i \left(\dot{Q}_i(t) \hat{e}_i(t) + Q_i(t) \frac{d\hat{e}_i}{dt} \right) \\ &= \frac{d\underline{Q}}{dt} \Big|_S + \underline{\omega} \times \underline{Q} \end{aligned}$$

$\underline{\omega} = \dot{\varphi} \hat{e}_3$

NB: 1. forces only on \underline{Q} not \underline{Q}_S or \underline{P}_{S_0}

$$2. \frac{d\underline{\omega}}{dt} \Big|_S = \frac{d\underline{\omega}}{dt} \Big|_{S_1} + \underline{\omega} \times \underline{\omega}$$

$\Rightarrow \underline{\omega} = \text{constant in one frame} \rightarrow \text{constant in rotated frame.}$

+ the derivations of $\underline{\omega}$ are the same as well.

Newton's eq in a rotating frame

$$m \underline{\ddot{r}} \Big|_{S_1} = \underline{F} \quad \text{in an inertial frame.}$$

$$\text{in } S: \quad \underline{\ddot{r}} \Big|_{S_0} = \underline{\ddot{r}} \Big|_S + \underline{\omega} \times \underline{r} \quad \text{from the above.}$$

$$\begin{aligned}
\ddot{\underline{r}}|_{S_0} &= \left(\frac{d}{dt} \Big|_S + \underline{\omega} \times \right) \dot{\underline{r}}|_{S_0} \\
&= \left(\frac{d}{dt} \Big|_S + \underline{\omega} \times \right) \left(\dot{\underline{r}}|_S + \underline{\omega} \times \underline{r} \right) \\
&= \frac{d}{dt} \dot{\underline{r}}|_S + \underline{\omega} \times \dot{\underline{r}}|_S + \underbrace{\frac{d}{dt} \Big|_S (\underline{\omega} \times \underline{r})}_{\underline{\omega} \times \dot{\underline{r}}|_S + \dot{\underline{\omega}} \times \underline{r}} \\
&\quad + \underline{\omega} \times (\underline{\omega} \times \underline{r}) \\
&= \ddot{\underline{r}}|_S + 2 \underline{\omega} \times \dot{\underline{r}}|_S + \dot{\underline{\omega}} \times \underline{r}|_S + \underline{\omega} \times (\underline{\omega} \times \underline{r})
\end{aligned}$$

↑
neglected

\underline{S}_A in the rotation frame (top the S)

$$m \ddot{\underline{r}} = \underline{F} - 2m \underline{\omega} \times \dot{\underline{r}} - m \underline{\omega} \times (\underline{\omega} \times \underline{r}) - m \dot{\underline{\omega}} \times \underline{r}$$

↑ Coriolis force
↑ centrifugal force
↑ azimuthal force

Centrifugal force:

first take $\underline{\omega} = \omega \hat{z}$, $\underline{r} \in x-y$ plane

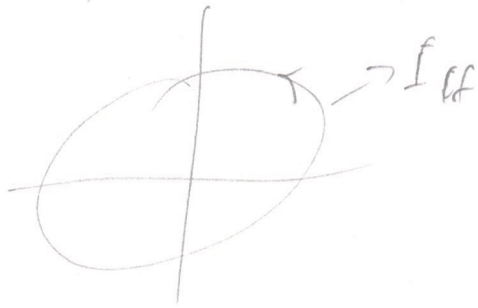
$$\underline{\omega} \times \underline{r} = \begin{vmatrix} \hat{x} & \hat{y} & \hat{z} \\ 0 & 0 & \omega \\ x & y & 0 \end{vmatrix} = -\omega y \hat{x} + \omega x \hat{y}$$

$$\underline{\omega} \times (\underline{\omega} \times \underline{r}) = \begin{vmatrix} \hat{x} & \hat{y} & \hat{z} \\ 0 & 0 & \omega \\ 0 & \omega x & -\omega y \end{vmatrix}$$

$$= \hat{x} (-\omega^2 x) + \hat{y} (-\omega^2 y) = -\omega^2 \underline{r}$$

or $\underline{F}_{cf} = -m \underline{\omega} \times (\underline{\omega} \times \underline{r}) = +m\omega^2 r \hat{r} = \text{outward force}$

Physically, think of riding in a car in a circle

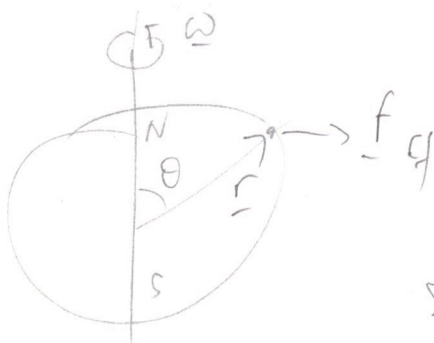


ground frame: you circle
acceleration = inward
inward force for ~~car~~ you

car frame: you're at rest but feel centrifugal
force pushing you against the door.

Same result (=force) in your.

Centrifugal force on Earth:



$$\underline{\omega} \times \underline{r} = \omega R \sin \theta \hat{n}$$

\hat{n} = into the plane

so $\underline{\omega} \times (\underline{\omega} \times \underline{r}) = \omega^2 R \sin \theta$ left

$-m \underline{\omega} \times (\underline{\omega} \times \underline{r}) = m\omega^2 R \sin \theta$ right

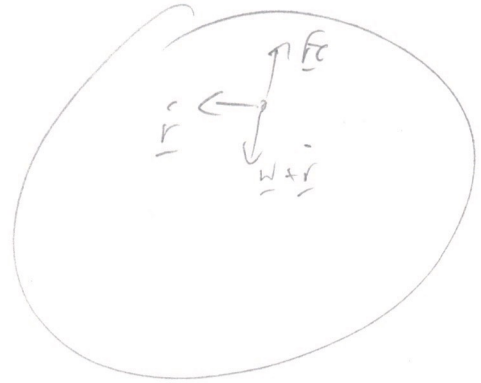
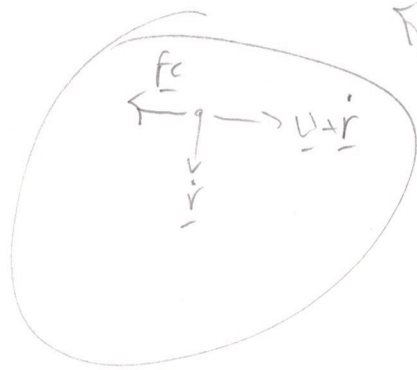
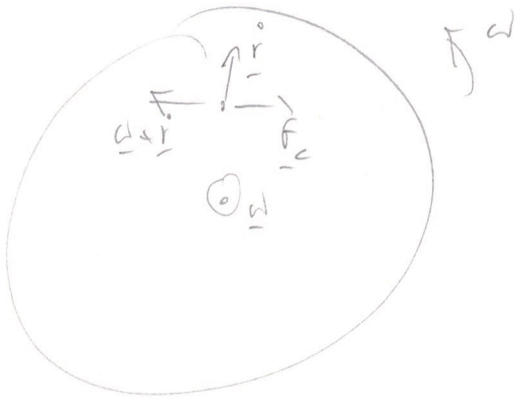
or $\underline{F}_{cf} = m\omega^2 R \sin \theta \hat{p}$

NB $\omega_{\text{Earth}} = \frac{2\pi}{1 \text{ day}} = 7.3 \times 10^{-5} \text{ s}^{-1}$, $R \sim 6370 \text{ km}$, $v^2 R \sim \omega^2 R^3$

Coriolis force

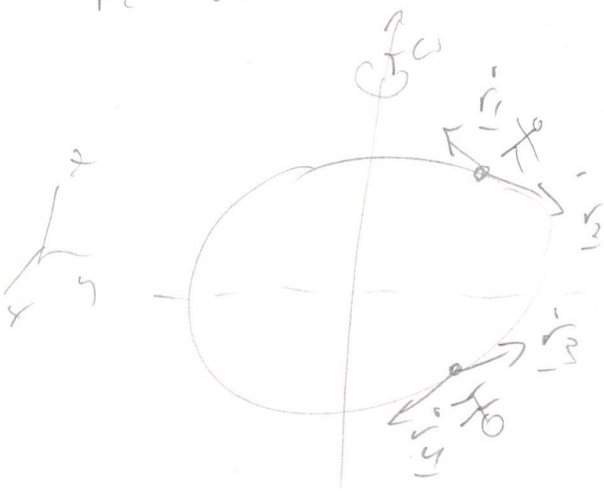
$$F_c = -2m \underline{\omega} \times \underline{r}$$

in a turntable:



force/deflection to right of direction of motion

in Earth



motion in xy plane because
in Northern Hemisphere

but in Southern you're "upside down"
+ right/left reversed

- deflection to left

or, direction of $\underline{\omega}$ reverses

for which N/S on Earth:

North
hem ($\underline{\omega} \times \underline{r}_1$ out of plane, $\underline{f}_{c,1}$ into plane
 $\underline{\omega} \times \underline{r}_2$ into plane, $\underline{f}_{c,2}$ out of plane) } deflect right

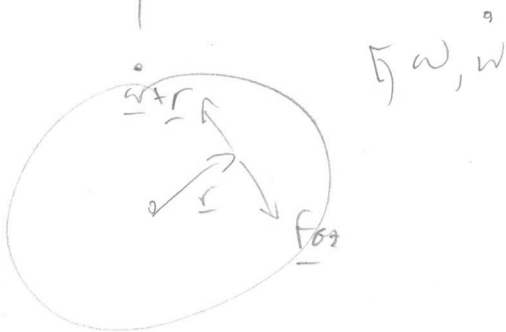
South
hem ($\underline{\omega} \times \underline{r}_3$ into plane, $\underline{f}_{c,3}$ out of plane
 $\underline{\omega} \times \underline{r}_4$ out of plane, $\underline{f}_{c,4}$ into plane) } deflect left

→ Coriolis hand swirl story -

* $\underline{f}_{oz} = -m \underline{\omega} \times \underline{r} = -m \underline{\omega} \hat{z} \times \underline{r}$
 (not often considered)

accelerating tangible

extra force pushing back
 in rotating frame
 → treat acc. case



* Note since $\omega \sim 10^{-4} \text{ sec}^{-1}$ need large \dot{r} to see an effect compared to $g = 10 \text{ m/s}^2$